

MAULANAABULKALAM AZAD UNIVERSITY OF **TECHNOLOGY, WEST BENGAL**

Paper Code: EC-801C

SATELLITE COMMUNICATION & REMOTE SENSING

Time Allotted: 3 Hours Full Marks: 70 The figures in the margin indicate full marks. Candidates are required to give their answers in their own words as far as practicable. http://www.makaut.com

Group - A

		(Multiple Choice Type	Quest	ions)		
1.	Choo	se the correct alternatives for the followings (any to	en):	1×10=10		
	(i)	A satellite in geostationary orbit may be describe the equator and whose orbit takes 24 hours.	d as a	satellite approximatelykm above		
		(a) 30,000	(b)	22,300		
		(c) 36,000	(d)	40,000		
	(ii)	INTELSAT stands for				
		(a) International Telecommunications Satellite	(b)	Indian Telecommunications Satellite		
		(c) Inter Telecommunications Satellite	(d)	None of these		
	(iii)	The refractive index of the ocean water				
		(a) increases with salinity.	(b)	increases with temperature.		
		(c) decreases with salinity.	(d)	decreases with temperature.		
	(iv)	The arrangement of terrain features which pro objects, is called http://www.makaut.com	vides	attributes : the shape, size and texture of		
		(a) spectral variation .	(b)	spatial variation		
		(c) temporal variation	-	none of these		
	(v)	The instruments which provide electromagnetic radiation of specified wavelength or a band of wavelengths to illuminate the earth surface, are called				
		(a) sensors		passive sensors		
		(c) active sensors		none of these		

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	(vi)	Thebasic requirement of any sensor system,						
	•	(a) radiometricresolution	-	spatial and spectral resolution .				
		(c) temporalresoulution	, ,	all of these				
	(vii)	The reflectance from a surface is called spec						
		(a) Snells's law		Lambert's cosine law				
		(c) Planktan's law	(d)	All of these				
	(viii)	For an elliptical orbit						
		(a) $e = 0$	•	1 < e > 0				
		(c) $e = 1$	(d)	None of the above				
	(ix)	Apogee is defined as						
		(a) the point farthest from earth.	(b)	the point nearest from earth.				
		(c) the point smallest from earth.	(d)	none of the above.				
	(x)	Which one is a Transfer orbit.						
		(a) LEO	(b)	Hohman -				
		(c) MEO	(년)	HAP				
	(xi)	The down link frequency in the C band trans	-					
		(a) 6 GHZ		4 GHz				
		(c) 14 GHz		11 GHz				
	(XII)	The carrier to noise ratio for a satellite depe		Bandwidth				
		(a) Effective Isotropic Radiated power						
		(c) Free space path losses		All of them				
	(xiii)	The signal from a satellite is normally aime						
		(a) path	(b)	effect				
		(c) footprint	· (d)	none of the above				
	(xiv)	The multiple access technique used for digi						
		(a) TDMA	(b)	FDMA				
		(c) Both (a) and (b)	(d)	None of them				
		Group						
		(Short Answer T	-	,	5×3=15			
	/	Answer any three of the following.						
2.	State	Kepler's three laws of planetary motion.			5			
3.	(a)	What do you mean by Sidereal day?			2			
	(b)	Explain the orbital parameters to determine t	he satellite	orbit	3			
4.	Prove	that the path of a satellite around the earth is	s an ellipse	and best described by :				
$\Gamma = \frac{p}{1 + \varepsilon \cos(\theta - \theta_0)}$								

where, $p = \frac{h^2}{\mu}$ and $e = \frac{h^2}{\mu}C$, they have their usual meanings.

5. Prove that the expression of velocity of a satellite is expressed as:

 $\vartheta^2 = \mu \left(\frac{2}{r} - \frac{1}{a}\right)$, where, r = radial distance a = semi-major axis and also find the values of perigee

 (ϑ_p) apogee (ϑ_a)

3+2=5

- 6. Describe Hohmann Orbital Transfer for a satellite and derive total velocity increment $(\theta_H = \Delta \theta_1 + \Delta \theta_2)$ in terms of standard parameters and also with neat diagram.

 3+2=5
- 7. (a) What do you mean by Powered Flight?
 - (b) With necessary schematic diagram derive the Rocket equation.

2+3=5

Group - C

(Long Answer Type Questions)

Answer any three of the following.

15×3=45

5

- 8. (a) What do you mean by Two Body problem in Orbital mechanics? Use necessary diagram.
 - (b) Derive the expression of mean anomaly (M) as known as Kepler's Equation described below:
 5
 M = E e sin E, where, E = eccentric anomaly and e = eccentricity.
 - (c) Prove that required change in velocity due to change in longitude is given by:

 (Δv) total = $\frac{2}{3}r \dot{\lambda}$, where $\dot{\lambda}$ = drift rate and r = radial distance

- 9. (a) Using Friss Transmission equation derive that Free space Loss (L_{FS}) is given by $L_{FS} = 22 + 20 \log_{10} \left(\frac{d}{\lambda}\right) dB \text{ for, general satellite communication link.}$
 - (b) What do you mean by Noise Power?

2

(c) Prove that, Carrier to Noise Ratio (C/N) for a satellite communication is expressed as:

 $\frac{C}{N} = \frac{P_T C_T G_R \left(\frac{\lambda}{4\pi d}\right)^2}{K T_S B L_A}, \text{ using standard parameters.}$

(d) What do you mean by Geostationary Orbit?

3

5

10. (a) Derive the complete link design for satellite communication given as follows:

 $(C/N_0)_T^{-1} = (C/N_0)_u^{-1} + (C/N_0)_D^{-1}$

- (b) For an earth station having an antenna of diameter 30 m and overall efficiency of 68%, the working signal frequency is 4150 MHz. At this frequency the system noise temperature is 79K when the antenna points at the satellite at an elevation angle of 28°. Calculate G/T ratio of the earth station. In case the sky noise temperature rises to 88 K, what would be the new G/T ratio?
- (c) Describe the Telemetry, Tracking and Command facility of a Satellite Communication System.

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11.	(a)	What is attitude control system? How it performs its operation?	5		
	(b)	Draw the block diagram of a Transponder and explain its stages.	5		
	(c)	What are different multiple access scheme used in satellite communication? Make a costudy.	mparative 5		
12.	(a)	Define sub-satellite point.	2		
	(b)	A satellite is an elliptical orbit with a perigee of 1000 km and apogee 4000 km. Using mean earth radius of 6378.14 km, find the orbital period in hours.			
	(c)	What is Orbital perturbation? Explain briefly.	3		
	(d)	A satellite at a distance of 36,000 km form the surface of the earth radiates a power of 57 antenna of gain 16dB. Find the flux density and power received by an antenna of eff 14 sq.m. If the receiving antenna has a gain of 60dB, calculate the received power.	W, from an ective area		
13.	(a)	Describe how sea surface temperature can be measured in remote sensing?	4		
	(b)	Explain how TRMM Microwave Imager (TMI) works to provide rainfall information?	5		
	(c)	Explain the spectral and radiometric resolution.	4		
	(d)	Distinguish between active and passive remote sensing.	2		
J.A.	Wri	te short notes on any three:	3×5=15		
	(a)	Radio Ocultation			

- (b) Radiometer
- (c) Code Division Multiple Access
- (d) GPS system
- (e) RADAR Imaging